



# High Power Density Carbon Neutral Electrical Power Generation for Air Vehicles

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### **Project Vision**

"We are solving the electrification of aviation by integrating the propulsion, power, and thermal systems for an energy optimized aircraft."



Range Extenders for Electric

Aviation with Low Carbon and

High Efficiency (REEACH)

REEACH / ASCEND / CABLES Annual Program Review Meeting June 28-30, 2022













### **Team**

Team member	Location	Role in project	
Rory Roberts, Tennessee Tech University	Cookeville, TN	Project Lead, System design lead -18 years experience in SOFC technology - Integrated propulsion, power, thermal expert	
Ted Ohrn, Special Power Sources	Alliance, OH	SOFC tube and stack design-Lead -30 years experience in SOFC's	SPS Special Power Sources Supplying Your REMOTE POWER NEEDS
Roland Dixon, Special Power Sources	Alliance, OH	Technology Transfer and Outreach-Lead -40 years government PM experience	SPS Special Power Sources Surpelving Your REMOTE POWER NEEDS
Chuck Lents, Raytheon Technologies	East Hartford, CT	Turbo-machinery & generator – Lead • 35 years in aircraft integrated system • RTRC electrified propulsion research lead	Raytheon Technologies
Kashif Nawaz, Oak Ridge National Lab	Oak Ridge, TN	Fluid-thermal modeling and design -Expert in high temperature thermal mng.	OAK RIDGE National Laboratory
John Hull, Boeing	Seattle, WA	Integration with aircraft	Ø BOEING
Mitch Wolff, Wright State University	Dayton, OH	Turbogenerator integration -Jet engine expert	WRIGHT STATE UNIVERSITY



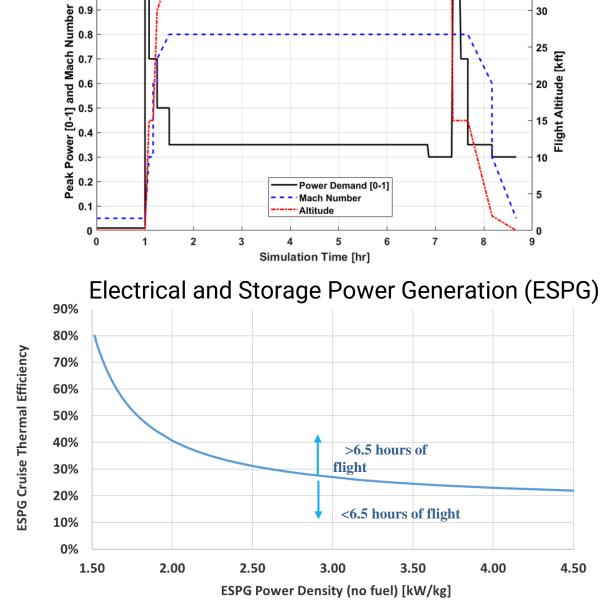
# Large Scale Electric Propulsion Approach

- Requires high conversion efficiency to drive down:
  - \$/passenger mile
  - Net zero Emissions
  - Meet current fuel storage requirements
- Vehicle level top-down design approach is required
- Reliability in approach for aerospace standards
- High power density electric centric systems:
  - Electric power production (REEACH)
  - Electric propulsors (ASCEND)
  - Electrical distribution system (CABLE)
- Manage 100's kWs of thermal management



### **REEACH ESPG Requirements**

- 28.7 MW Take-off Electrical Power
- 8.5 MW Cruise Power
- ▼ > 3 kW-hr/kg
- < <0.15 \$/kWh
- ▼ ESPG< 25,804 kg (includes fuel)</p>
- ▼ 6.5 hours Flight time
- Ambient Temperature range, -54° to 30°C
- Ambient Pressure range, 23 kPa to 101 kPa
- **Initial Targets** 
  - ESPG> 1.12 kW/kg
  - SOFC-C-TG > 1.6 kW/kg
  - At 65% ESPG Efficiency



**Commercial Flight Profile** 

⊲ 35

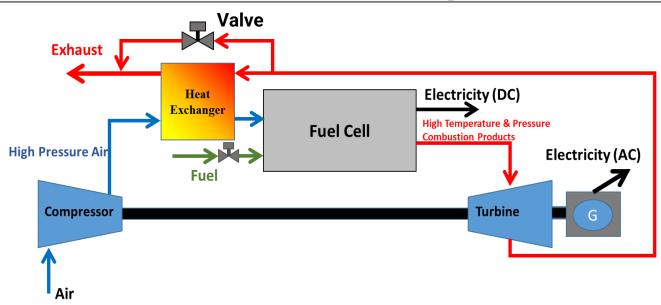
30

Flight Altitude

4.50



# Conventional Fuel Cell-Gas Turbine Hybrid (FC-GT)



### **Pros**

- FC-GT provides ultra high chemical-to-electrical conversion efficiency
- Provides pressurized environment at high altitudes

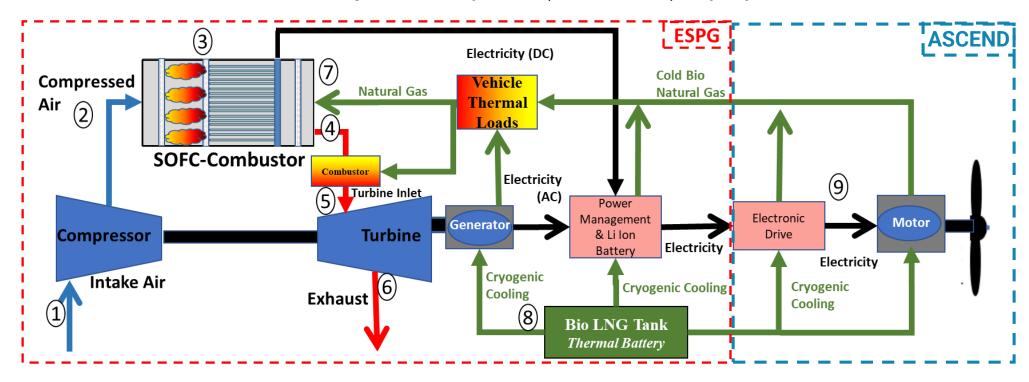
### **Cons**

- Large massive systems with low specific power
- Large thermal mass, sluggish response to perturbations
- Long cold startup times
- Complex thermal management of fuel cell typically with large valves



### **Proposed Integrated ESPG Concept**

Solid Oxide Fuel Cell Combustor-Turbogenerator system (SOFC-C-TG) is proposed for the ESPG



#### **Pros**

- SOFC-C-TG provides a simple & elegant solution for electric power generation in air vehicles
- SOFC-C-TG eliminates cathode heat exchangers, large thermal mass. *Minimum size and weight*
- Provides precise thermal control of SOFC stack at cathode inlet. Minimum use of valves
- Rapid response to perturbations and extreme conditions: load, inlet temperature and pressure.
- Redundancy and reliability

#### Cons

New concept, never been fully demonstrated

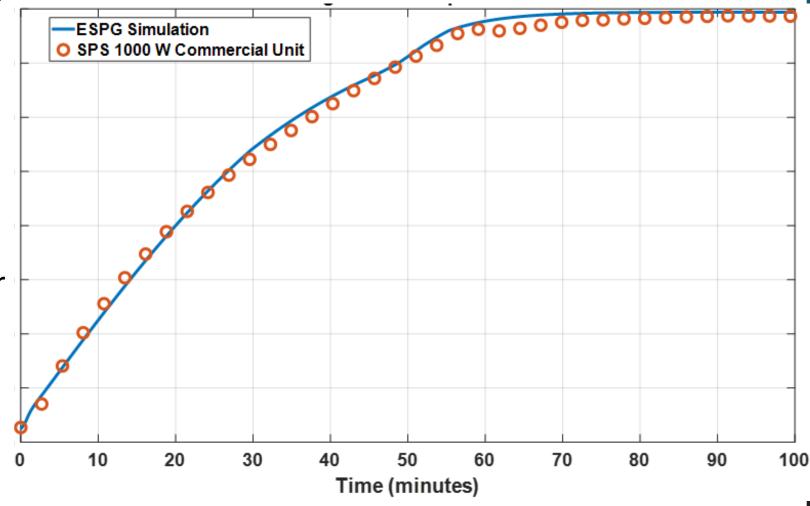
### Vehicle Level Transient Analysis

- ESPG model integrated with a notional 737 Boeing class aircraft
  - Objective: To establish detailed boundary conditions and required operational characteristics of the ESPG components
  - MATLAB/Simulink detailed transient model with vehicle level controls
    - Quasi-2D SOFC: electrochemistry overpotentials, reformation and electrochemical kinetics, heat transfer included
    - TG includes: performance maps, shaft dynamics, unsteady flow
    - Combustors: combustion reaction and products, heat transfer
    - Electrical Power Conditioning System: Vehicle electrical load balance and control
    - Fuel Thermal Management System: bio LNG tanks, fuel heat exchangers, valves

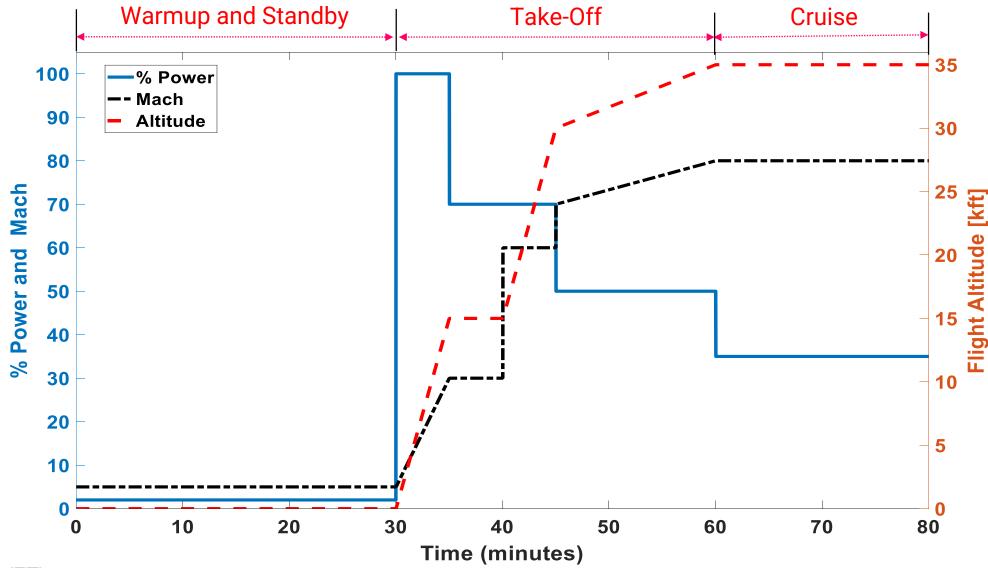
Results are preliminary, sizing parameters and vehicle controls need further refinement

# Simulated Warmup Compared to Measured Data

- ▼ The ESPG dynamic model was modified to account for corrected air flow and SOFC mass to parametrically match a 1,000 W commercial Special Power Sources SOFC unit.
- ▼ The dynamic model was able to predict the warmup sequence.
- ▼ 60 minutes was required for this warmup, but < 20 minutes has been demonstrated with commercial units in the field.

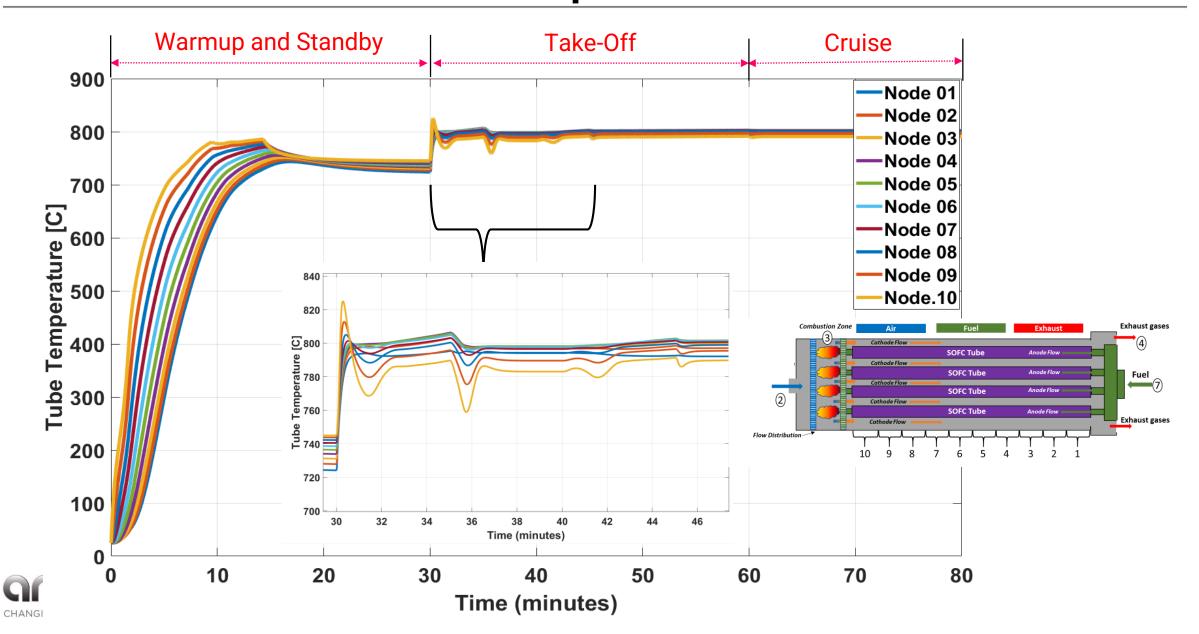


# **Notional Flight-Simulation**

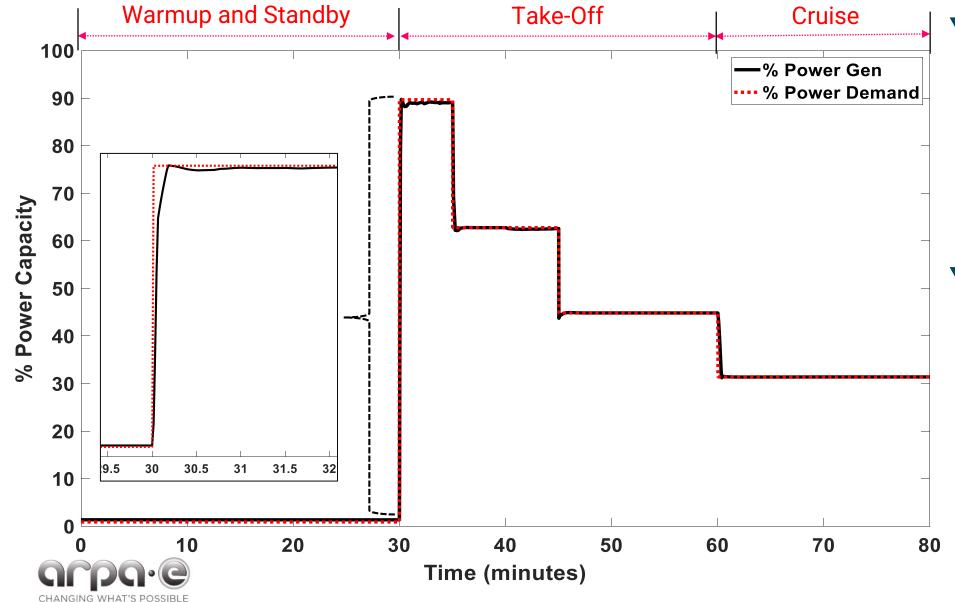




# Simulated SOFC Stack Warmup

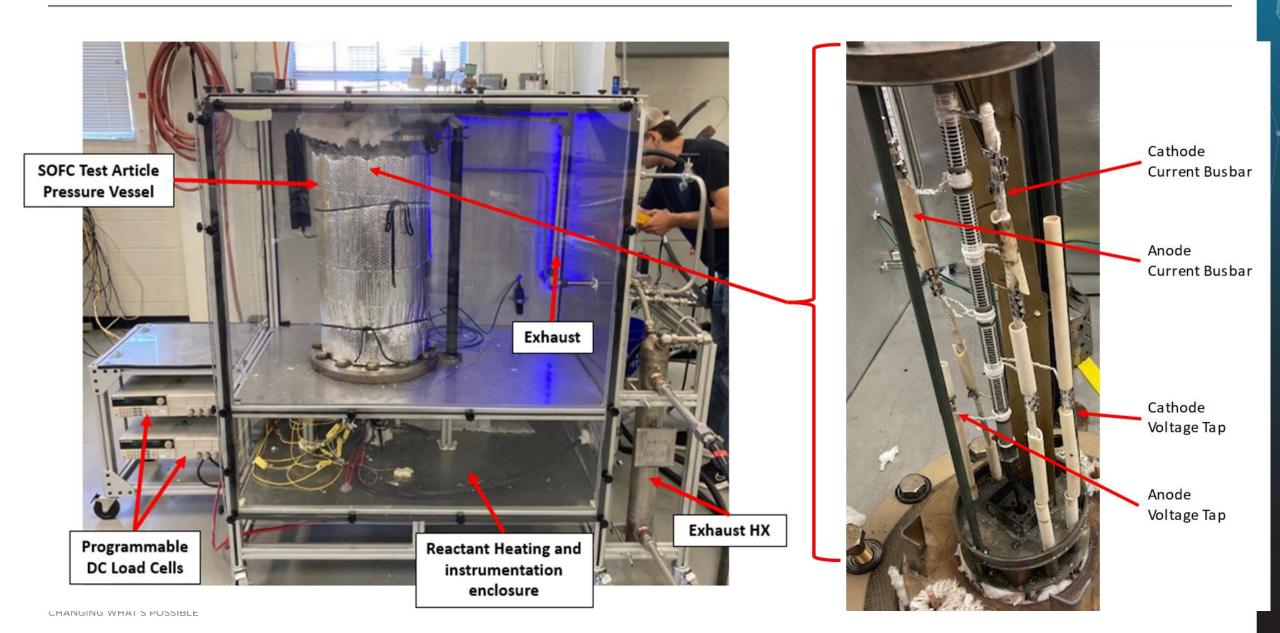


### Simulated ESPG Power Load Follow

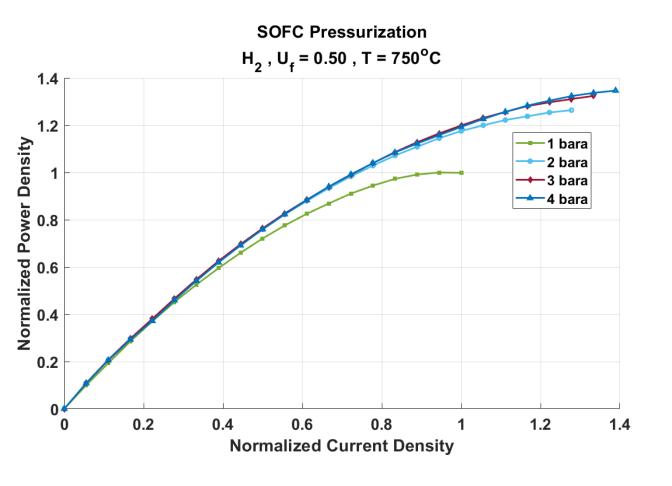


- ▼ The ESPG system was able to meet the dynamic vehicle electrical load demands, 90% step change in 1 minute
- Small battery storage (100 kWhr) would be required to meet the large step change in power demand (worst case)

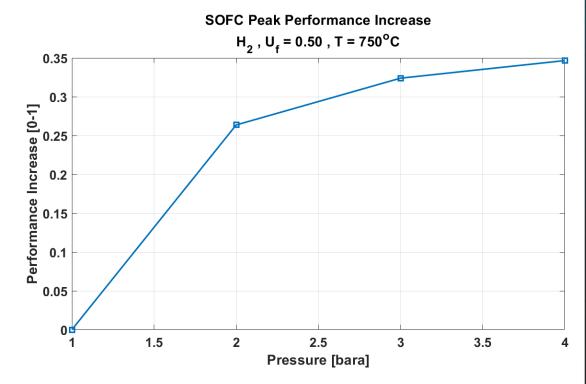
# **SOFC Pressurized Testing**



### **Anode Supported SOFC Pressurization**

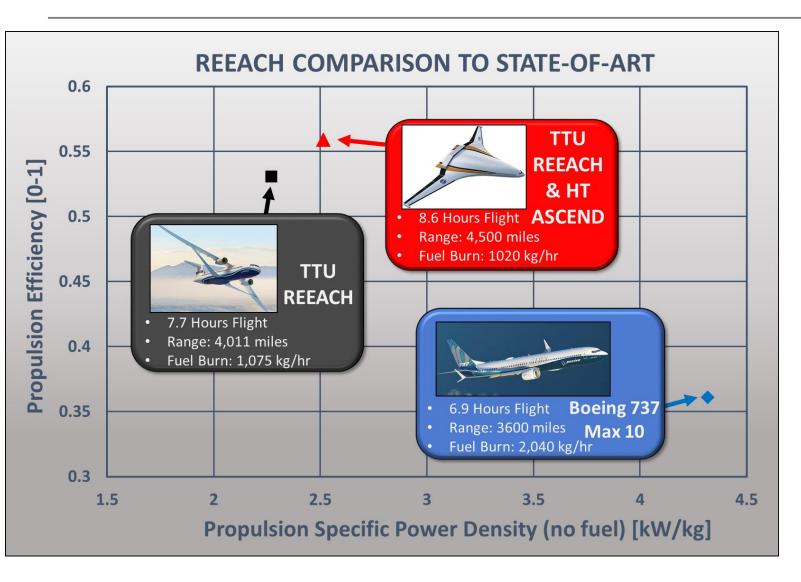


- ▼ SOFC performance was normalized using the peak ambient power density and current density
- ▼ 35% increase from 1 to 4 bara





# Summary of REEACH Comparison State-of-Art



### Assumptions:

- Power Gen. Thermal efficiency 65%
- Power Gen. Specific Power 3.6 kW/kg
- Bio LNG for REEACH, Jet A for 737 Max

#### TTU REEACH

- Motor drive specific power -127 kW/kg
- Motor specific power 12 kW/kg
- Motor efficiency 93%

#### TTU REEACH & HT ASCEND

- Motor drive specific power -127 kW/kg
- Motor specific power 20 kW/kg
- Motor efficiency 98%



### Task Outline & Technical Objectives

- Phase 1 Objective: Demonstrate 1 kW SOFC-C operating within the boundary conditions of a ESPG designed to exceed state-of-art propulsion
- Task 1: Task and Milestone Negotiation/Project management
- Task 2: ESPG System Design
- Task 3: Single Tube Development
- Task 4: Multi-Tube SOFC-C Bundle Development
- Task 5: Technology to Market
- Phase 2 Objective: Demonstrate 5 kW SOFC-C operating within the boundary conditions of a ESPG designed to exceed state-of-art propulsion



July 8, 2022

# **Risk Update**

Risk	#
Development of high-power density low weight cells	1
Pressurized operation of the SOFC stack	2
On-anode reforming within SOFC tubes	3
Combustor instability for large range of equivalence ratios and residence times	4
SOFC-C integration with TG	5
SOFC-C transformative power technology and limited technical expertise exists to adopt for wide commercial use	6

	Almost					
Likelihood	Certain					
	Likely			(6)		
				6		
	Moderate		3		1	
	oue.ute			1 5		
	Unlikely		4 4	5 2	2	
			3			
	Rare		3			
	Naic					
		Insignificant	Minor	Moderate	Major	Catastrophic
	Consequences					



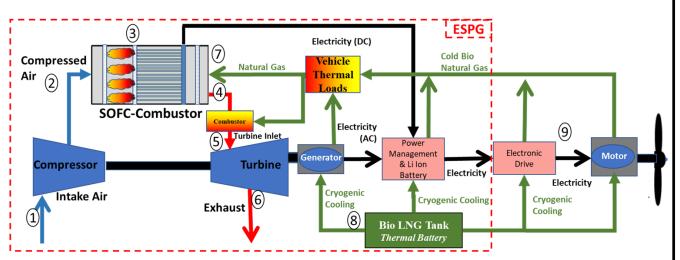








### REEACH 2240-1541 T2M Commercialization Plan – Jun 2022



### **Team Plan to Commercialization**

- Planned Business Model is internal business unit with SPS lead
- Business Plan drafted and being internally reviewed
- Freedom to operate analysis underway

#### **Anticipated First Markets:**

#### Product/Service

1) SOFC-C or EPSG Configuration

#### Description

- 1) UAV Group 4 i.e. MQ-1
- 2) Regional Aircraft <75 seat Electric Aircraft
- Light Air Cargo UPS/medical products and military

#### **Pricing Strategy**

 SOFC-C or EPSG Configuration – Fixed Price PO with margins set by market conditions

#### **Anticipated Long-Term Markets**

#### **Product/Service**

1) SOFC-C or EPSG Configuration

#### **Description**

1) Commercial Aircraft – 186 seat Electric Aircraft

#### **Pricing Strategy**

 SOFC-C or EPSG Configuration – Fixed Price PO with margins set by market conditions

#### Anticipated Services/Spares

#### Product/Service

 SOFC-C or EPSG Configuration Service Contracts and Spares

#### **Description**

1) Commercial Aircraft , Regional and Light Air Cargo

#### **Pricing Strategy**

 SOFC-C or EPSG Configuration – Regional Service & Spares Distribution Centers



1st Market Dates Long-Term Markets Services/Spares

NLT 2030 NLT 2032 NLT 2030

# **Needs and Potential Partnerships**

- Needs project has currently:
  - Contact with airline customer for establishing customer needs
- Anticipated needs following the completion of the award:
  - Application of RTX generator capability to develop a multi-megawatt machine, not being addressed in this program
  - Power electronics to integrate and regulate voltage and current from SOFC and generator.
  - Infrastructure expertise related to aviation flight support
- Capabilities that could be useful for other REEACH teams:
  - Integrated propulsion, power, and thermal management expertise
  - Pressurized SOFC operation expertise
  - LNG for aviation fuel expertise



# **Q & A**





https://arpa-e.energy.gov

